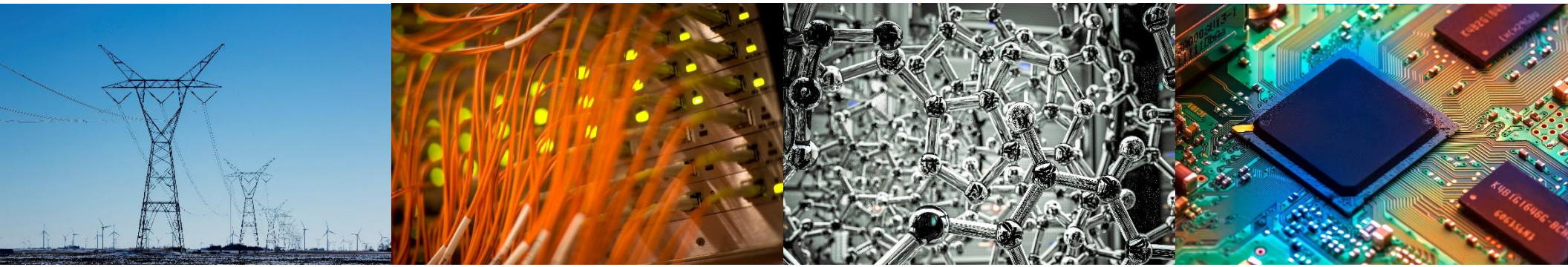


# DIRECT DESIGN OPTIMIZATION FOR TRAJECTORY PERFORMANCE

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# Design Minimality as Optimization

minimize  
«*design params*»

«*resource usage*»

subject to

«*design feasibility*»

«*laws of physics*»

# Direct Numerical Design Optimization

- Computers are great at solving optimization problems!
- Meeting input requirements of numerical solvers can be intractable
  - Requires real/discrete valued cost functions, decision variables, and constraints.
  - **Robot functionality and task structure would need to be specified implicitly or explicitly as constraints on the design parameters**

# Minimality for Task Performance

- Consider subset of design optimization primarily concerned with task kinematics and dynamics

minimize  
«*design params*»

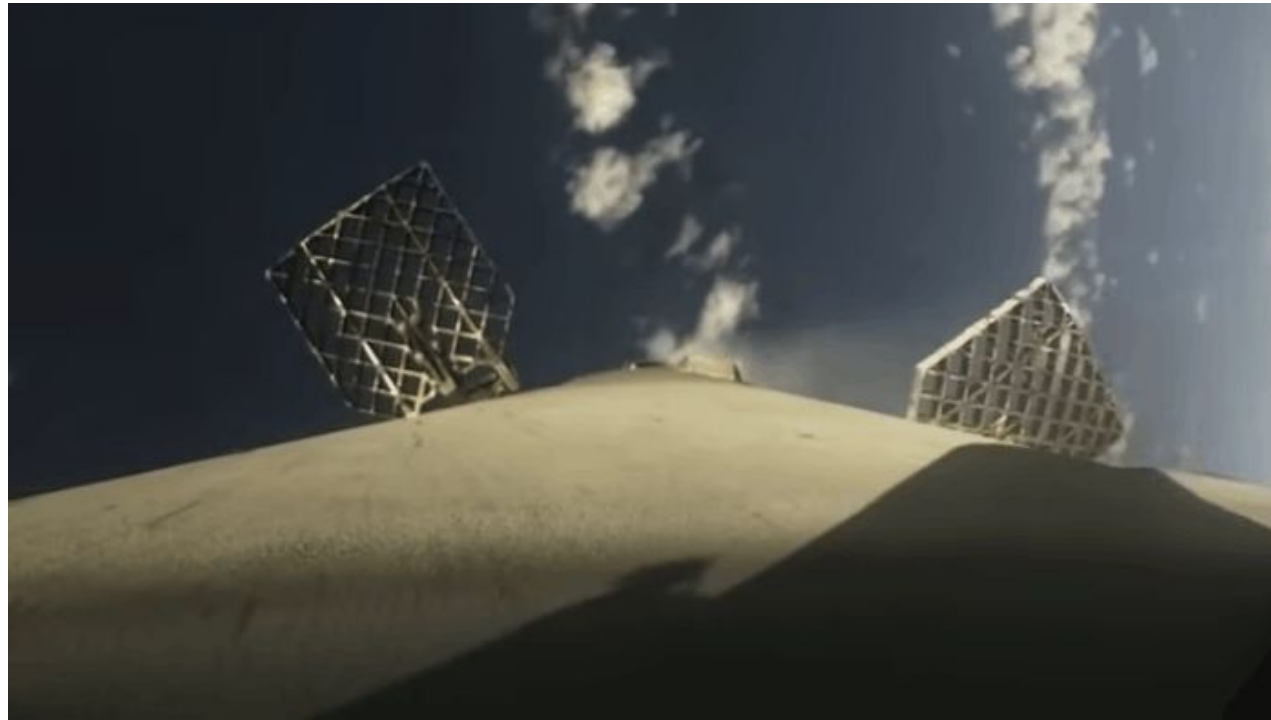
subject to

«*task cost + design cost*»

«***task** feasibility*»

«*laws of physics*»

# Example: Falcon 9



minimize

«*design params*»

«*price per launch*»

subject to

«*orbit reachability*»

«*rigid body dynamics, aerodynamics, etc.*»

Image: SpaceX

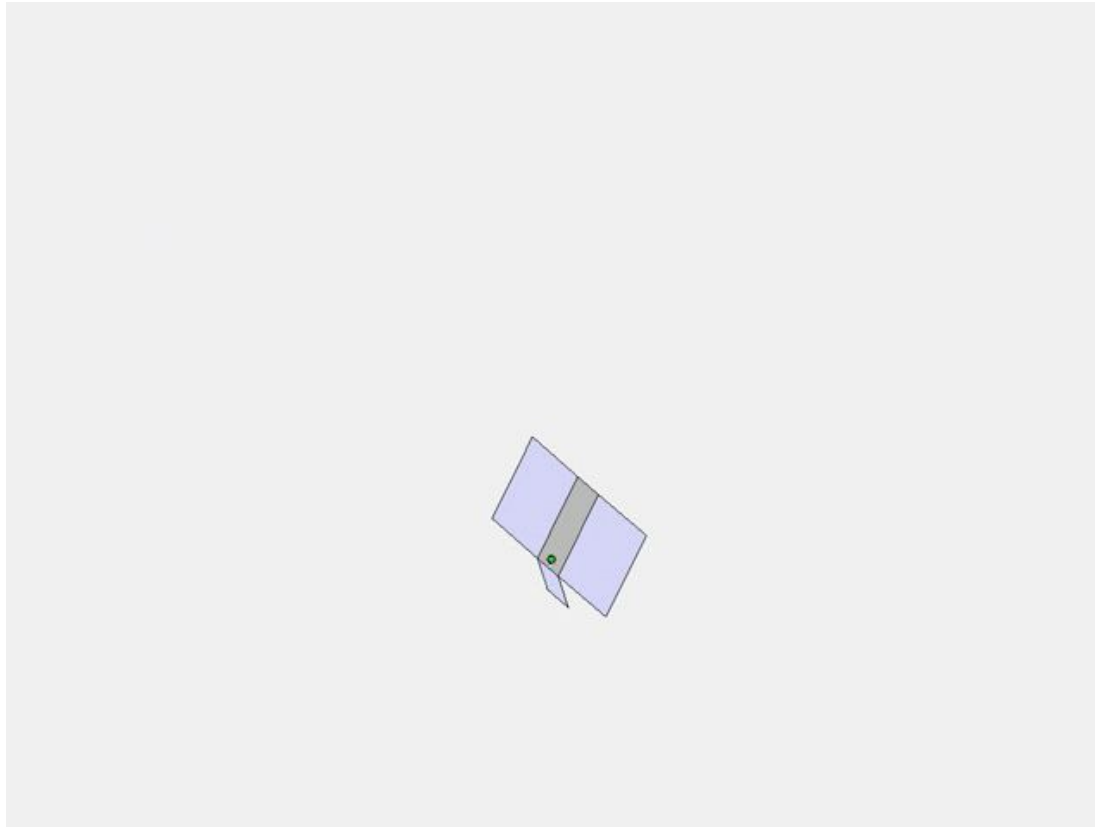
# Trajectory Optimization

- Idea: Create process for design optimization by leveraging computational tools used for solving similar problems
- *Trajectory Optimization* problems search for input  $\mathbf{u}$  that optimizes some *cost-to-go*  $J$  and satisfies constraints  $\mathbf{d}$  for a dynamical system  $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u})$

$$\begin{array}{ll} \text{minimize} & \int_{t_0}^{t_f} l(\mathbf{x}(t), \mathbf{u}(t)) dt + J_f(\mathbf{x}(t_f)) \quad // \text{resource usage / cost} \\ \mathbf{u}(\cdot) & \\ \text{subject to} & \dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}(t), \mathbf{u}(t)), \forall t \in [t_0, t_f] \quad // \text{laws of physics} \\ & \mathbf{d}(\mathbf{x}(t), \mathbf{u}(t)) \geq \mathbf{0}, \forall t \in [t_0, t_f] \quad // \text{trajectory feasibility} \end{array}$$

# Trajectory Optimization Examples

- Multi-contact traversal of uneven terrain (Dai, 2016)
- Dexterous hand manipulation (Mordatch, 2012)
- Perching of flapping-wing systems (Halm, 2017)



# Direct Trajectory Optimization

- Strategy: solve direct approximation of continuous time problem by optimizing over a sampling of the inputs and state
- *Direct Transcription* explicitly enumerates samples of state and input trajectory as decision variables

$$\begin{aligned} & \underset{(\mathbf{x}_0, \mathbf{u}_0), \dots, (\mathbf{x}_N, \mathbf{u}_N)}{\text{minimize}} && \sum_{i=1}^N \frac{\Delta t_i}{2} (l(\mathbf{x}_{i-1}, \mathbf{u}_{i-1}) + l(\mathbf{x}_i, \mathbf{u}_i)) + J_f(\mathbf{x}_N) && // \text{resource usage / cost} \\ & \text{subject to} && \Delta t_i > 0, \forall i \in 1, \dots, N \\ & && \mathbf{g}(\Delta t_i, \mathbf{x}_{i-1}, \mathbf{u}_{i-1}, \mathbf{x}_i, \mathbf{u}_i) = \mathbf{0}, \forall i \in 1, \dots, N && // \text{laws of physics} \\ & && \mathbf{d}(\mathbf{x}_i, \mathbf{u}_i) \geq \mathbf{0}, \forall i \in 0, \dots, N && // \text{trajectory feasibility} \end{aligned}$$

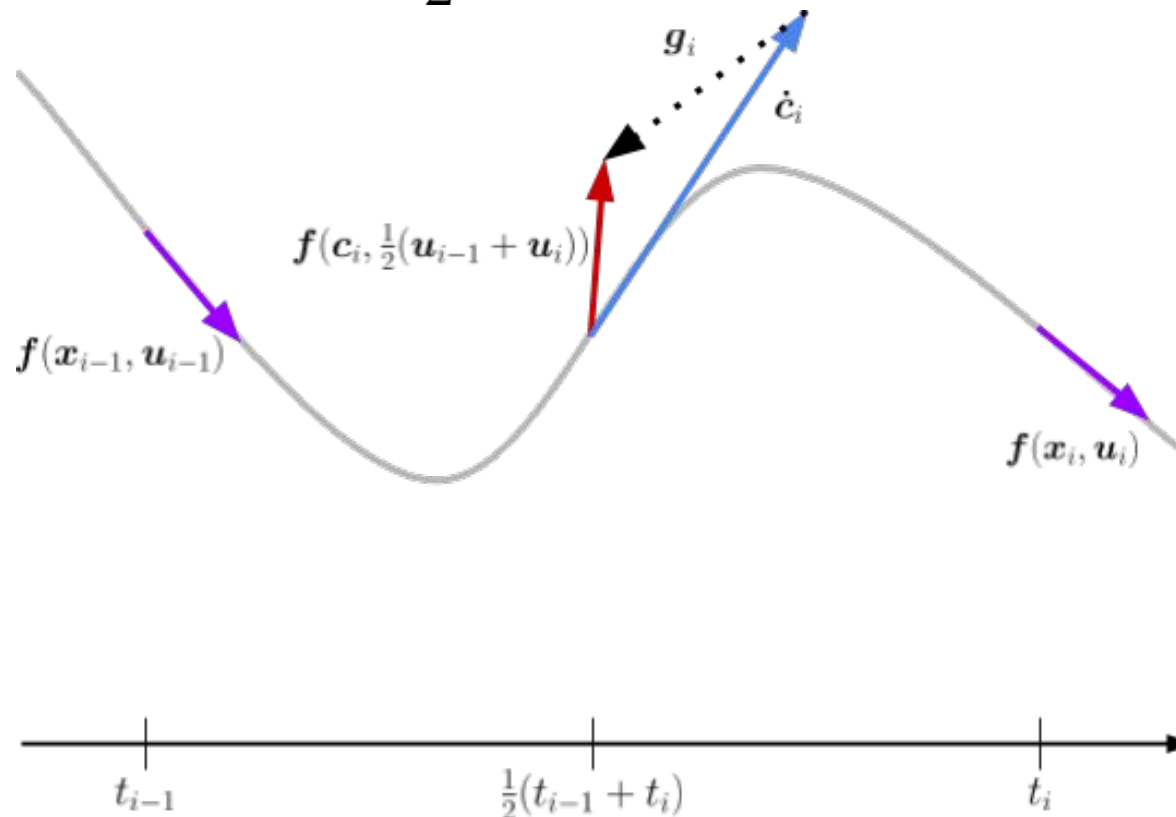


# Direct Trajectory Optimization

$$\mathbf{g}_{euler} = \mathbf{x}_i - (\mathbf{x}_{i-1} + \Delta t_i \mathbf{f}(\mathbf{x}_{i-1}, \mathbf{u}_{i-1})) = 0$$

$$\mathbf{g}_{back} = \mathbf{x}_i - (\mathbf{x}_{i-1} + \Delta t_i \mathbf{f}(\mathbf{x}_i, \mathbf{u}_i)) = 0$$

$$\mathbf{g}_{hermite} = \mathbf{f}(\mathbf{c}_i, \frac{1}{2}(\mathbf{u}_{i-1} + \mathbf{u}_i)) - \dot{\mathbf{c}}_i = 0$$



# Direct Design Optimization

- Add design parameters  $\alpha$  explicitly as decision variable to optimization problem
- Handle variations in equations of motion  $\mathbf{f}$ , trajectory costs  $l$  and  $J_f$  and trajectory constraints  $\mathbf{d}$  by adding dependence on  $\alpha$
- Add additional, independent design cost  $J_d$ , and augment  $\mathbf{d}$  with any necessary design feasibility constraints.
- Solution of resulting problem should provide optimal task trajectory and optimal system parameters.

$$\underset{(\mathbf{x}_0, \mathbf{u}_0), \dots, (\mathbf{x}_N, \mathbf{u}_N), \alpha}{\text{minimize}} \quad \sum_{i=1}^N \frac{\Delta t_i}{2} (l(\mathbf{x}_{i-1}, \mathbf{u}_{i-1}, \alpha) + l(\mathbf{x}_i, \mathbf{u}_i, \alpha)) + J_f(\mathbf{x}_N, \alpha) + J_d(\alpha)$$

subject to

$$\Delta t_i > 0, \forall i \in 1, \dots, N$$

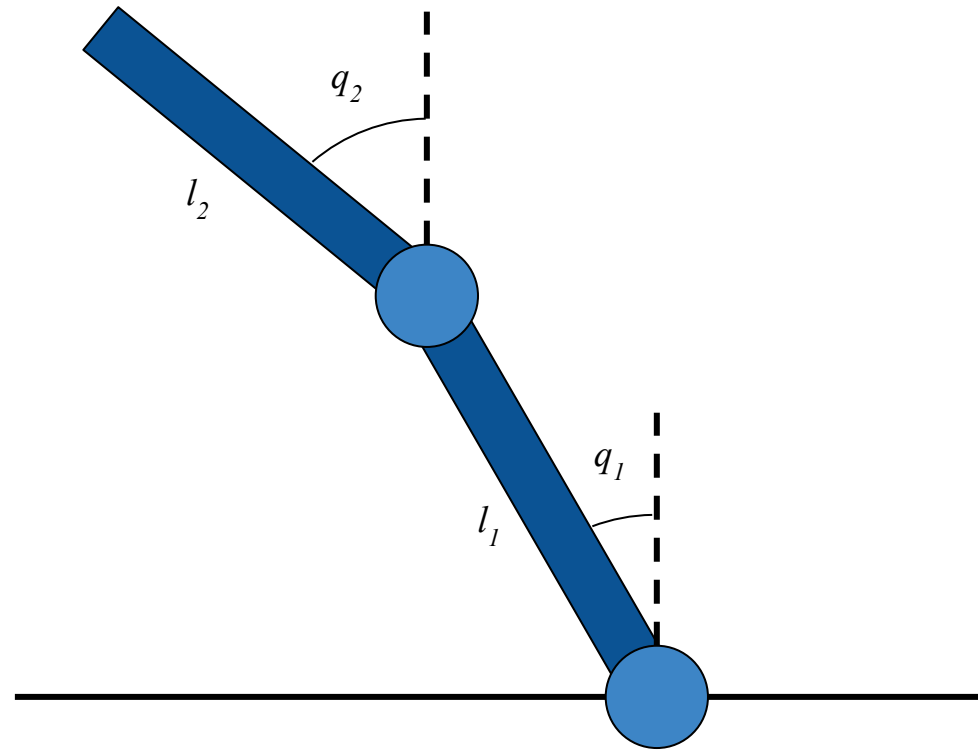
$$\mathbf{g}(\Delta t_i, \mathbf{x}_{i-1}, \mathbf{u}_{i-1}, \mathbf{x}_i, \mathbf{u}_i, \alpha) = \mathbf{0}, \forall i \in 1, \dots, N$$

$$\mathbf{d}(\mathbf{x}_i, \mathbf{u}_i, \alpha) \geq \mathbf{0}, \forall i \in 0, \dots, N$$

# Example: 2-Link Planar Arm

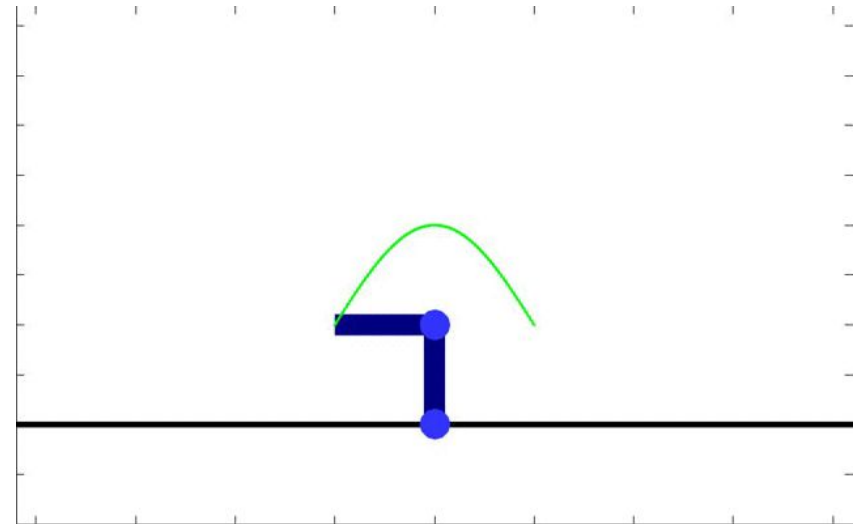
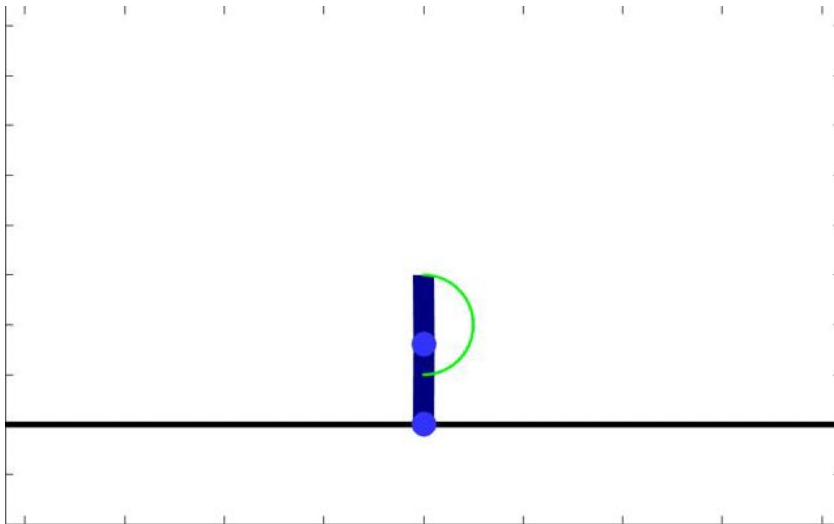
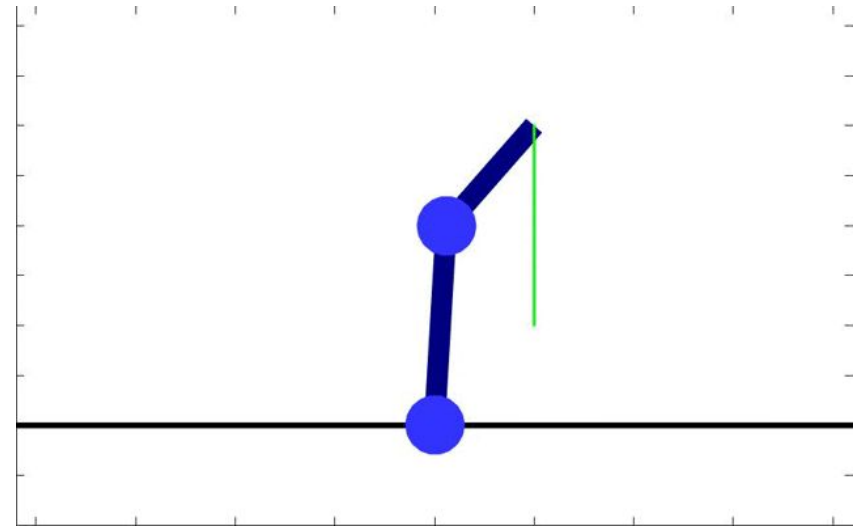
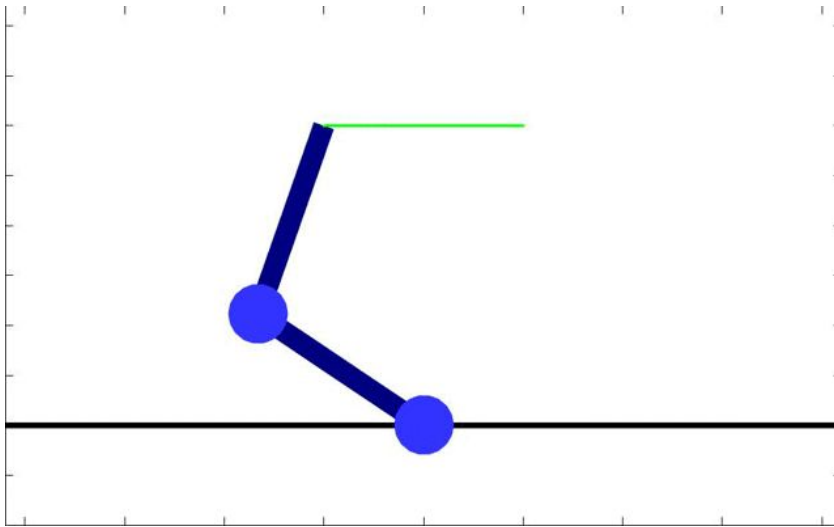
$$\mathbf{x} = \begin{bmatrix} \mathbf{q} \\ \dot{\mathbf{q}} \end{bmatrix}$$

$$\boldsymbol{\alpha} = \begin{bmatrix} l_1 \\ l_2 \end{bmatrix}$$



$$\dot{\mathbf{x}} = \begin{bmatrix} \dot{\mathbf{q}} \\ \mathbf{H}(\mathbf{q}, \boldsymbol{\alpha})^{-1} (\mathbf{B}\mathbf{u} - \mathbf{N}(\mathbf{q}, \dot{\mathbf{q}}, \boldsymbol{\alpha})) \end{bmatrix}$$

# 2-Link Planar Arm Results



# Discussion and Future Work

- Base formulation is incredibly flexible
  - Arbitrary nonlinear dynamics allow for application in a very wide range of scenarios, from UAVs to legged locomotion
  - Arbitrary constraints on design parameters allow for discrete decision variables (such as number of linkages) to be handled with Mixed Integer Programming
- Exact solutions to arbitrary problems is NP-Hard
  - High flexibility limits computational efficiency by ignoring structure
  - Only very small numbers of discrete variables can be handles for nonconvex problems
- Consider permuted formulations for large classes of robots, e.g. rigid body systems with frictional contact

# Q&A